SOLENT UNIVERSITY

SOUTHAMPTON

Reducing HAWT Blade Failure due to Fatigue Through Varied Aerodynamic Design

Introduction The project aimed to investigate the effects of changing the geometric properties of HAWT blade aerofoils on the aerodynamic forces. Specifically targeting those which contribute to fatigue without contributing to power output. Minimising these forces should help reduce the likelihood of blade failure by fatigue, while 80.00 maintaining a high overall turbine power coefficient. ist [kN] 65.00 S815 - Thick, High Camber. S827 - Thick, Medium Camber. 50.00 35.00 S826 - Thin, High Camber. S829 - Thin, Low Camber.

Project Method

- Iterations were used to calculate values for chord and Reynolds number at several points across the span.
- Chords were plotted to give design dimensions which were optimised for the mean lift coefficient of the 4 aerofoils.
- XFOIL was used to simulate air flow around the aerofoils and compute coefficients.
- Typical wind conditions were simulated based on the reference turbines wind class: II_A. $Re = 4.1 \times 10^6$
- Blade was divided into sections and Blade Element Momentum Theory was used to calculate the aerodynamic forces at each section.

Assumptions

• BEM assumes 2D air flow and that the flow in one BEM section does not affect neighbouring sections.

C1 -9.24 m

 Tangential and Axial Induction Factors assume initial conditions.

